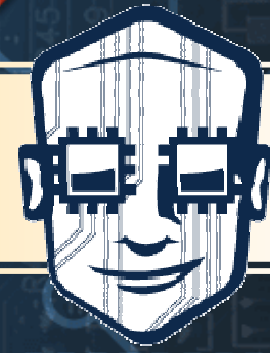


# **DESIGNCON<sup>®</sup> 2013**

**JANUARY 28-31, 2013**

**SANTA CLARA CONVENTION CENTER**



## **Reliability Modeling of Electronics for Co-Designed System Applications**

**DfR Solutions**  
reliability designed, reliability delivered



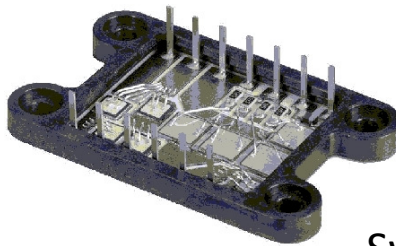
# Agenda

- Introduction
- High Reliability Applications
- Common Issues
- Lifetime Expectations
- Failure Mechanisms
- Virtual Qualification Approach
- Automated Design Analysis Solution

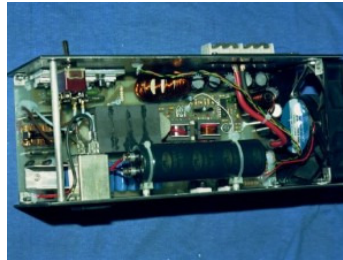
# High Reliability Systems are Prevalent in Several Market Segments



Automotive Power Modules



Switching Power Supply



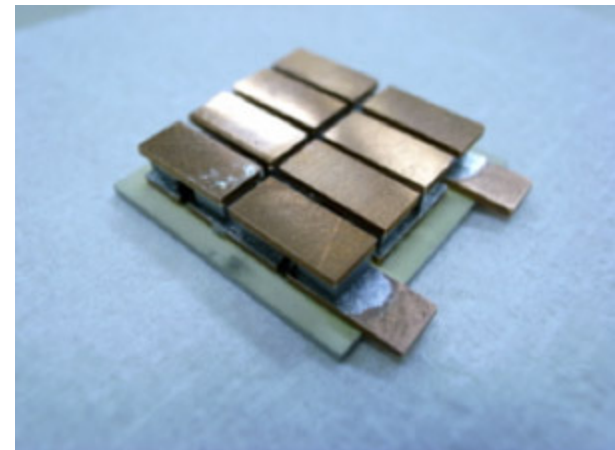
Solar Power Modules



Street Lighting



IGBT



Thermoelectric Modules



# What Do they All Have in Common?

- High Temperature Environments
- Possible Vibration and Shock Environments
- Temperature and Power Cycling Environments
- Very High Current Flows and Thermal Transfer Requirements
- A variety of materials forming the product
  - Substrate tiles bonded to copper baseplate



# Stringent Environmental Conditions

- Being used at varying temperatures or temperature extremes
- Having a temperature range of -55°C to 125°C
- Being used in an application having a medium to high shock, pressure, vibration, or moisture environment
- Being stored for later usage (over 10 years)
- Having an application life span of 10 to 25 years

# Example Life Expectancies

- IGBT – Rail application – 30 years (Each module 100FIT)
- Power Module – Automotive Application – 20 years
  - 10W/cm<sup>2</sup>
  - DBC Substrate bonded to heatsink
  - Vibration, shock, humidity, salt spray
  - Cost
- Solar Power Inverters-25 years
- Street Lighting – 15-20 years

# Failure Mechanisms

- Thermo-mechanical fatigue induced failures
  - CTE mismatch
  - Temperature swings
- Bond Wire Fatigue
  - Shear Stresses between bond pad and wire
  - Repeated flexure of the wire
  - Lift off (fast temperature cycling effect)
  - Heel Cracking
- Die Attach Fatigue
- Solder Fatigue
  - Voids
- Device Burn Out
- Automotive- degradation of power
  - Solder Fatigue
  - Bond wire failure (lift off due to fast temperature cycling)
- Structural Integrity – ceramic substrate to heat sink in thermal cycling
- IGBTs – solder joint fatigue, wirebond liftoff, substrate fracture, conductor delamination



# How Can We Resolve these Issues During the Design Phase of a Product?

- Utilize an Automated Design Analysis Approach Because:
  - Mil-HBK-217 actuarial in nature
  - Physics based algorithms are too time consuming
  - Need to shorten NPI cycles and reduce costs
  - Increased computing power
  - Better way to communicate

# PoF: the Complexity Roadblock

$$\tau_{HCI} \propto \exp\left[\frac{b_{HCI}}{V_D}\right] \cdot \exp\left[\frac{E_{aHCI}}{kT}\right]$$

$$L = L_r \left(\frac{V_r}{V_0}\right) \times 2^{\left(\frac{T_r - T_A}{10}\right)}$$

$$T_f \propto \exp\left(\frac{\sim 0.51eV}{kT}\right) \times \exp(\sim -0.063\%RH)$$

$$\tau_{TDDB} \propto \exp[-b_{TDDB} \cdot V_G] \cdot \exp\left[\frac{E_{aTDDB}}{kT}\right]$$

$$N_f^{-0.6} D_f^{0.75} + 0.9 \frac{S_u}{E} \left[ \frac{\exp(D_f)}{0.36} \right]^{0.1785 \log \frac{10^5}{N_f}} - \Delta \varepsilon = 0$$

$$\frac{t_1}{t_2} = \left(\frac{V_2}{V_1}\right)^n \exp \frac{E_a}{K_B} \left(\frac{1}{T_1} - \frac{1}{T_2}\right)$$

$$\tau_{EM} \propto (J)^{-n} \cdot \exp\left[\frac{E_{aEM}}{kT}\right]$$

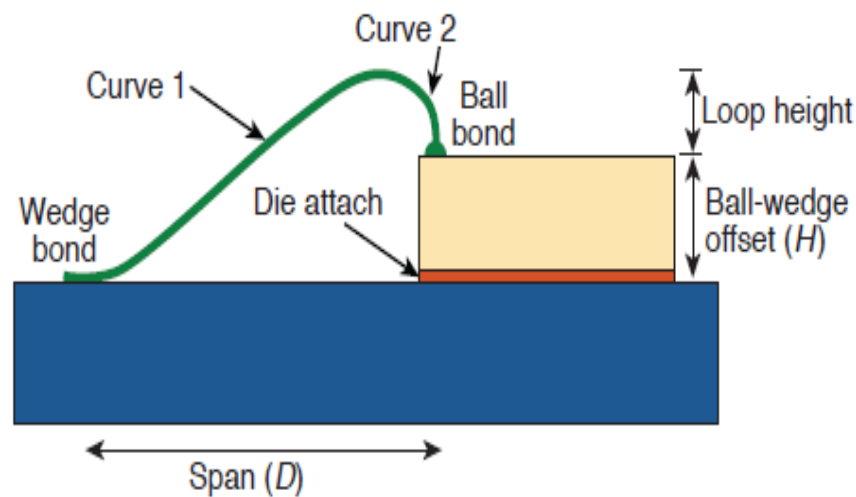
$$\tau_{NBTI} \propto \exp[-b_{NBTI} \cdot V_G] \cdot \exp\left[\frac{E_{aNBTI}}{kT}\right]$$

$$(\alpha_2 - \alpha_1) \cdot \Delta T \cdot L = F \cdot \left( \frac{L}{E_1 A_1} + \frac{L}{E_2 A_2} + \frac{h_s}{A_s G_s} + \frac{h_c}{A_c G_c} + \left( \frac{2 - \nu}{9 \cdot G_b a} \right) \right)$$

# Common Failure Modes

- Wire Bonds

Wire bonding has been the most common interconnect for IC packages for over 50 years. The most common materials are gold, aluminum, and more recently copper. The most common bond pad material is aluminum.



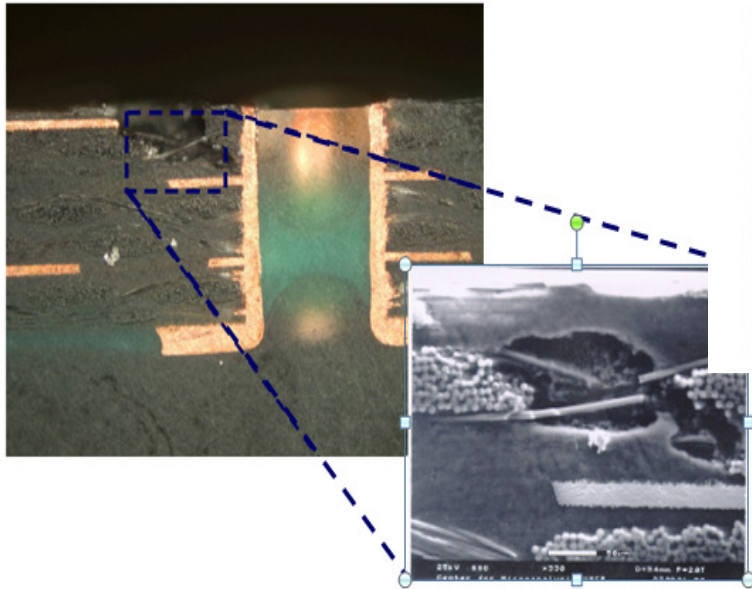
Wire bonds tend to fail if exposed to elevated temperatures (intermetallic formation), exposure to elevated temperature and humidity (corrosion) and exposure to temperature cycling (low cycle fatigue).



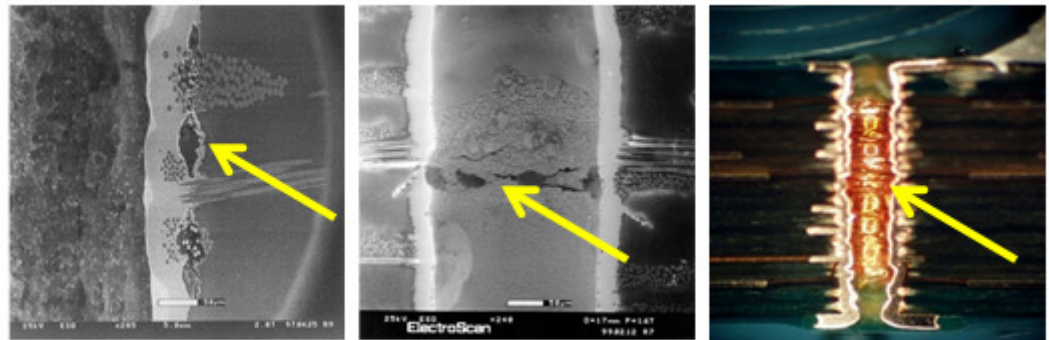
# Common Failure Modes: PCBs

- Printed Wiring Boards have several failure modes that are detrimental to reliable operation. Failures in PCBs can be driven by:
- Size (larger boards tend to experience higher temperatures)
- Thickness (thicker boards experience more thermal stress)
- Material (lower  $T_g$  tends to be more susceptible)
- Design (higher density, higher aspect ratios)
- Number of reflow exposures

# Common Failure Modes: PCBs



Conductive anodic filament (CAF), also referred to as metallic electromigration, is an electrochemical process which involves the transport (usually ionic) of a metal across a nonmetallic medium under the influence of an applied electric field. CAF can cause current leakage, intermittent electrical shorts and dielectric breakdown between conductors in printed wiring boards.



Plated Through Hole Failure Mechanisms: voids (left), etch pits (center) and barrel cracking from fatigue (right)

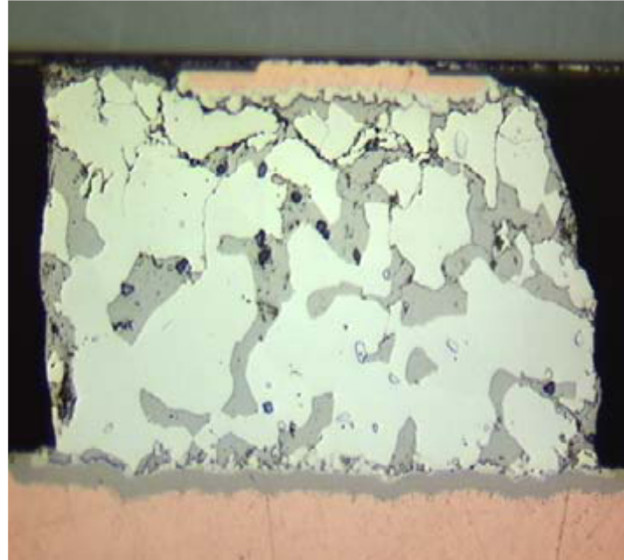
PTH voids can cause large stress concentrations, resulting in crack initiation.

Etch pits are due to either insufficient tin resist deposition or improper outer-layer etching process and rework.

Overstress cracking can occur in the PTH due to a Coefficient of Thermal Expansion (CTE) mismatch which places the PTH in compression.

# Common Failure Modes: Solder Fatigue

- Thermo-Mechanical Fatigue of solder joints is one of the primary wear-out mechanisms in electronic products. This is especially true in products used outside of commercial/ consumer environments where a longer lifetime is required and more severe operating conditions exist. The analysis assesses the fatigue of the solder joints as a function of the stresses applied during its lifetime and provides insight into whether joints are susceptible to failure.





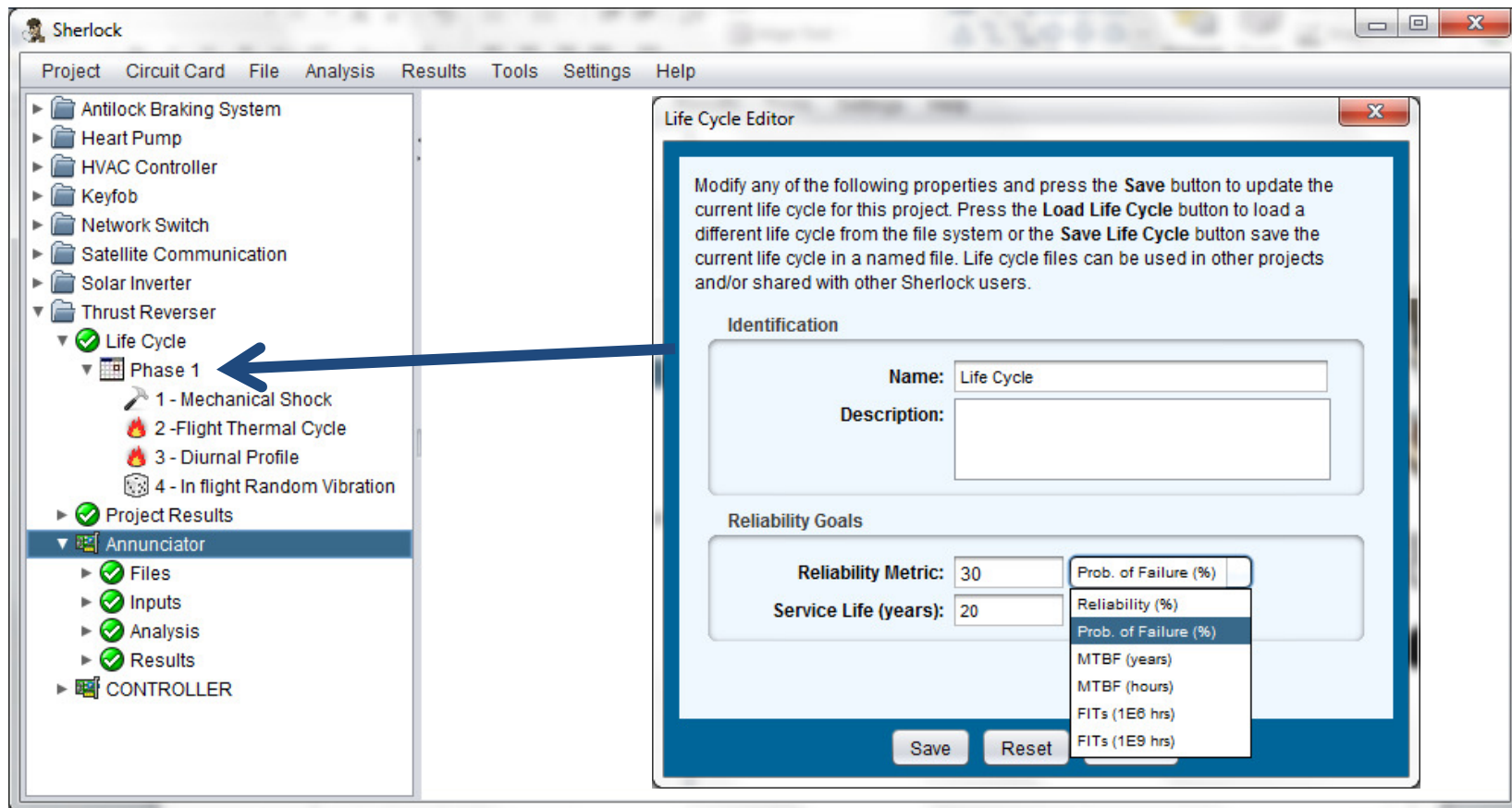
# Automated Design Analysis

- Easy to Utilize
- Easy to Locate commands
- Industry Terminology
  - Parts List
  - Stack-up
  - Pick and Place
  - ODB++
  - GERBER

There are several high levels steps involved in performing an automated design analysis. They are:

- Define Reliability Goals
- Define Environments
- Add Circuit Cards
- Import Files
- Generate Inputs
- Perform Analysis
- Interpret Results

# Reliability Goals



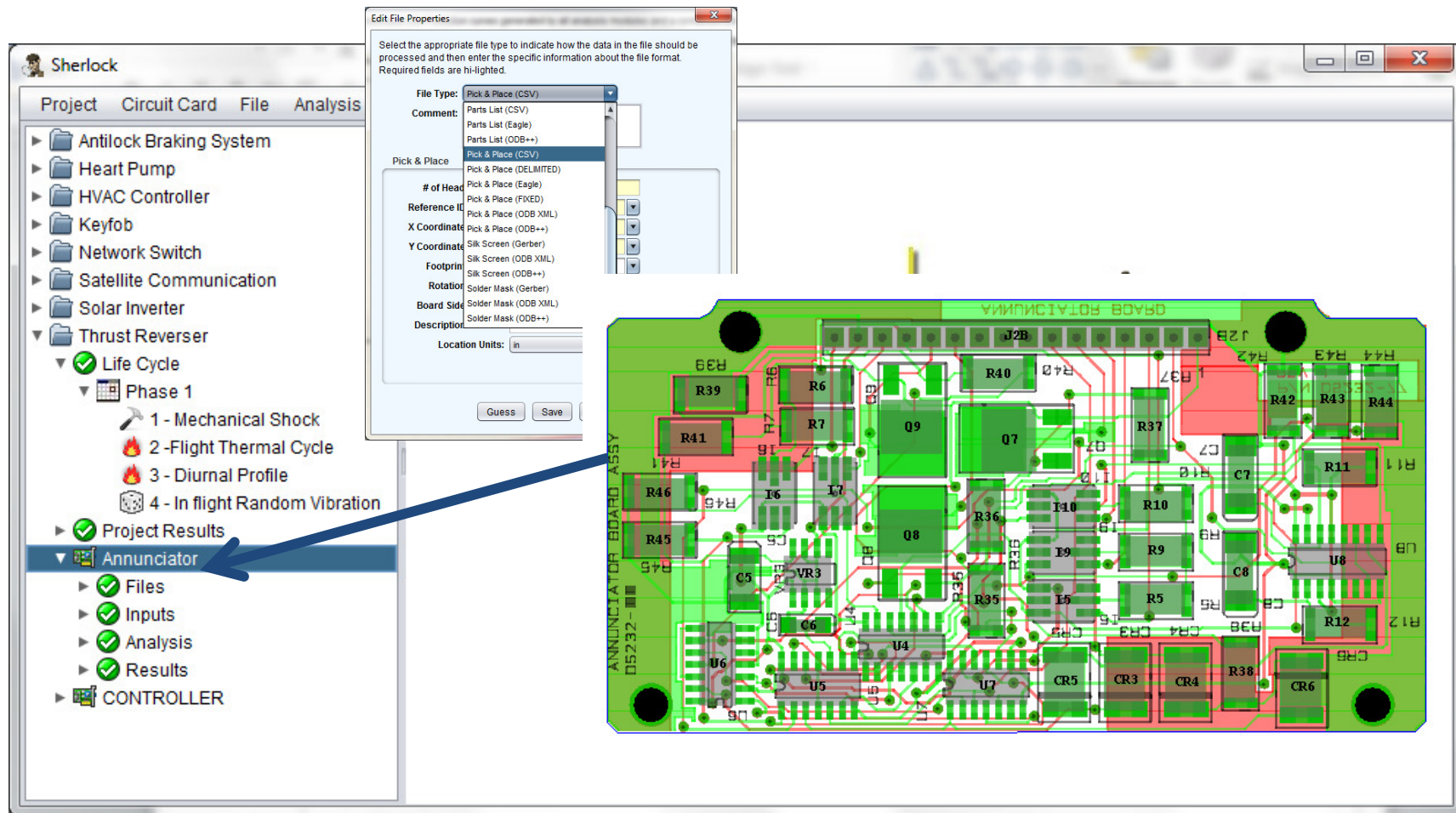
# Ambient Environment



Handles Very Complex Environments



# Input Design Files



# Input: Parts List

The screenshot displays the Sherlock software interface. The main window shows the 'Main Board Parts List' with a table of parts. The parts are color-coded by source: User (blue), PartsDB (orange), and BOM (yellow). The table includes columns for Ref Des, Part Number, and Part Type. Two 'Package Chooser' dialog boxes are open, showing the selection of a package for a specific part. The dialog boxes show a list of packages with columns for Package Type, Pin Count, Size (mm), and Package Name. The selected package is C-BEND-3528-21, which is a 2-pin, 3.2 x 2.8 mm package.

**Main Board Parts List**

Ref Des	Part Number	Part Type
U110	075-1313-5	IC
U700	075-0973-5	IC
U701	075-0973-5	IC
U702	075-0700-5	IC
U703	075-0700-5	IC
U704	075-0700-5	IC
U705	075-0700-5	IC
U707	075-0676-5	IC
U708	075-0676-5	IC
U709	075-0676-5	IC
U710	075-0676-5	IC
U720	075-0686-5	IC
U721	075-0686-5	IC
U722	075-0736-5	IC
U723	075-0099-5	IC
U724	075-1154-5	IC
U727	075-0560-5	IC
U728	075-0700-5	IC
U730	075-0667-5	IC
U731	075-0667-5	IC
U732	075-1154-5	IC
U733	075-0667-5	IC
U734	075-1386-5	IC
U738	075-0853-5	IC
Y2	076-0050-5	IC
Y3	076-0050-5	IC
Y4	076-0050-5	IC

**Package Chooser - U94**

The following properties are currently defined for the selected part as derived from the listed source. Press the ... button to see all source values for a given property.

Part Properties - U94

ID	Pkg	Thermal	Loc	Ball	Pad	Die	Flag	Lead	Qual
Ball Pattern:	?	FULL							comp-top.odb
Ball Count:	?	256							comp-top.odb
Ball Units:	?	mm							comp-top.odb
Ball Pitch:	?	1.0							comp-top.odb
Ball Diameter:	?	0.7							Guess
Ball Height:	?	0.6							Guess
Ball Chan Width:	?	0.0							comp-top.odb

**Package Chooser**

Select the desired package:

Package Type	Pin Count	Size (mm)	Package Name
ALL	ALL	ALL	QFN-4 (MO-220VEEB)
JEP95	4	1.0 x 1.0	QFN-6 (MO-252UAD)
BGA	6	1.4 x 1.0	QFN-6 (MO-252UAD)
CBEND	8	1.4 x 1.8	QFN-6 (MO-287UAF)
CC	10	1.4 x 2.0	QFN-6 (MO-287UFAD)
COIP	12	1.5 x 1.0	QFN-6 (MO-287X1AFA)
LCCC	14	1.5 x 1.5	QFN-6 (MO-287X1FAD)
LSOP	16	1.5 x 2.0	QFN-6 (MO-287X2FA)
MELF	18	1.6 x 1.2	QFN-6 (MO-287X2FA)
PDIP	20	1.6 x 1.6	QFN-8 (MO-208AAEA)
PDSO	22	1.6 x 2.1	QFN-8 (MO-220VEEC)
QFN	24	1.8 x 2.6	QFN-8 (MO-220VEEC-2)
QFP	28	2.0 x 1.0	QFN-8 (MO-220VEEC-3)
QFP / LQFP / TQFP	32	2.0 x 2.0	QFN-8 (MO-220VEEB)
	36	2.5 x 1.0	QFN-8 (MO-248UEEC-1)

Package Name: C-BEND-3528-21  
Package Type: CBEND  
Package Material: TANTALUM  
Package Leads: 2 - C-Lead - Copper  
Dimension (mm): 3.2 x 2.8 x 1.8

- Color coding of data origin
- Minimizes data entry through intelligent parsing and embedded package and materials database



# Inputs: Stack-Up

## Stackup Properties

The following board properties are based on the currently defined board outline and the individual layer properties shown below:

Board Size: 370 x 178 mm [14.6 x 7.0 in]      CTE<sub>xy</sub>: 17.519 ppm/C  
 Board Thickness: 1.6 mm [62.0 mil]      CTE<sub>z</sub>: 58.646 ppm/C  
 Board Density: 2.8515 g/cc      E<sub>xy</sub>: 35,743 MPa  
 Copper Layers: 10      E<sub>z</sub>: 4,379 MPa

## Stackup Layers

Double click any row to edit the properties for that layer or select one or more rows and press the **Edit Selected** button below to edit properties for a batch of layers. Press the **Generate Stackup Layers** button to replace all layers using a given PCB thickness and default layer properties.

Layer	Type	Material	Thickness	Density	CTE <sub>xy</sub>	CTE <sub>z</sub>	E <sub>xy</sub>	E <sub>z</sub>
1	POWER	COPPER (33.7%)	1.0 oz	4.1927	21.180	21.180	45,838	45,838
2	Laminate	Generic FR-4					24,804	3,450
3	POWER	COPPER (92.2%)					105,099	105,099
4	Laminate	Generic FR-4					24,804	3,450
5	POWER	COPPER (50%)					62,350	62,350
6	Laminate	Generic FR-4					24,804	3,450
7	POWER	COPPER (92.2%)					105,099	105,099
8	Laminate	Generic FR-4					24,804	3,450
9	POWER	COPPER (91.1%)					103,984	103,984
10	Laminate	Generic FR-4					24,804	3,450
11	POWER	COPPER (19.9%)					31,859	31,859
12	Laminate	Generic FR-4					24,804	3,450
13	POWER	COPPER (92.2%)					105,099	105,099
14	Laminate	Generic FR-4					24,804	3,450
15	POWER	COPPER (23.2%)	1.0 oz	3.4472	21.747	21.747	35,202	35,202
16	Laminate	Generic FR-4	5.3 mil	1.9000	17.000	70.000	24,804	3,450
17	POWER	COPPER (92.2%)	1.0 oz	8.3462	18.021	18.021	105,099	105,099
18	Laminate	Generic FR-4	5.3 mil	1.9000	17.000	70.000	24,804	3,450
19	POWER	COPPER (20.9%)	1.0 oz	3.2839	21.871	21.871	32,872	32,872

**Edit Selected Layers**

Enter any fields to be updated in all selected layers. If a field is left blank, then the current field value will remain unchanged for each layer.

**Laminate Layer Properties**

Laminate Material:

Laminate Thickness:

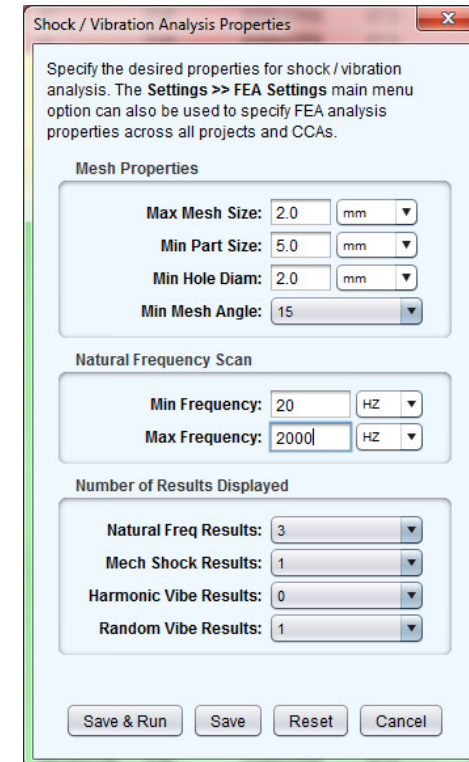
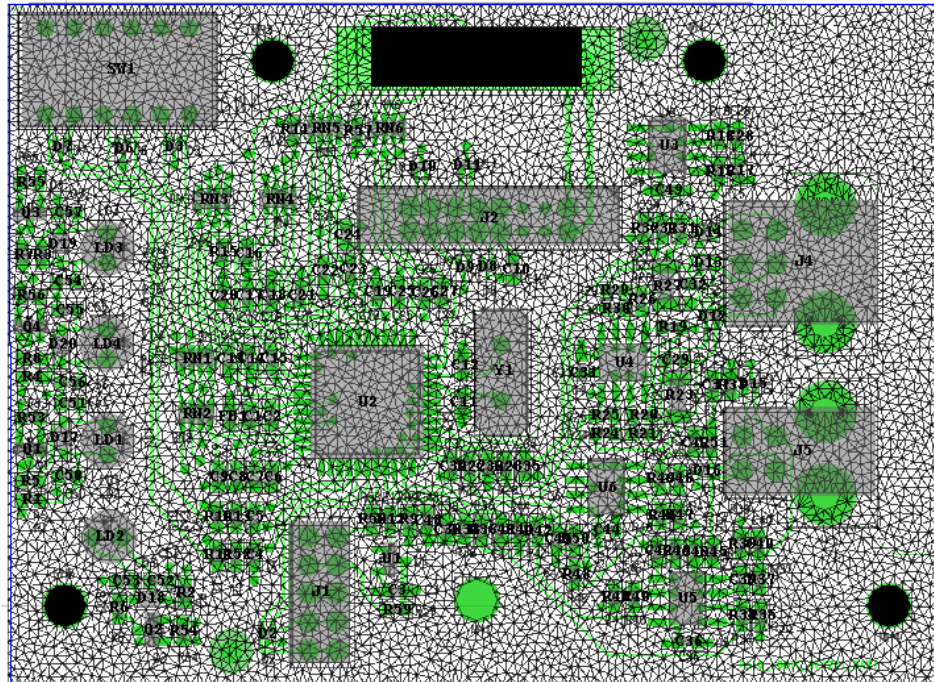
- Automatically generates stackup and copper percent (%)
- Embedded database with almost 400 laminate materials with 48 different properties

# Analyses

- Eight different analyses can be performed. They are:
  - CAF – Conductive Anodic Filament Formation
  - Plated Through Hole Fatigue
  - Solder Joint Fatigue
  - Finite Element Simulations
  - ICT Impact
  - DFMEA
  - Vibration Fatigue - Natural Frequencies
  - Mechanical Shock



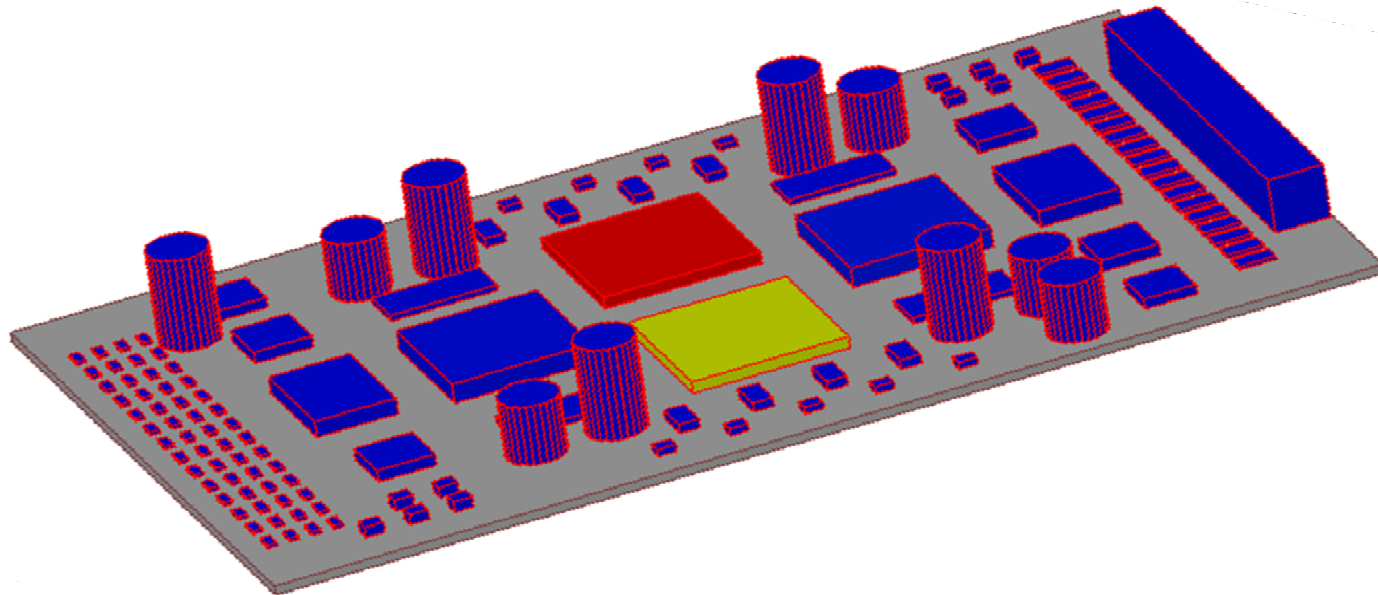
# Results: Automated Mesh Generation



- Identifies optimum mesh density based on board size
- Expert user no longer required; model time reduced by 90%

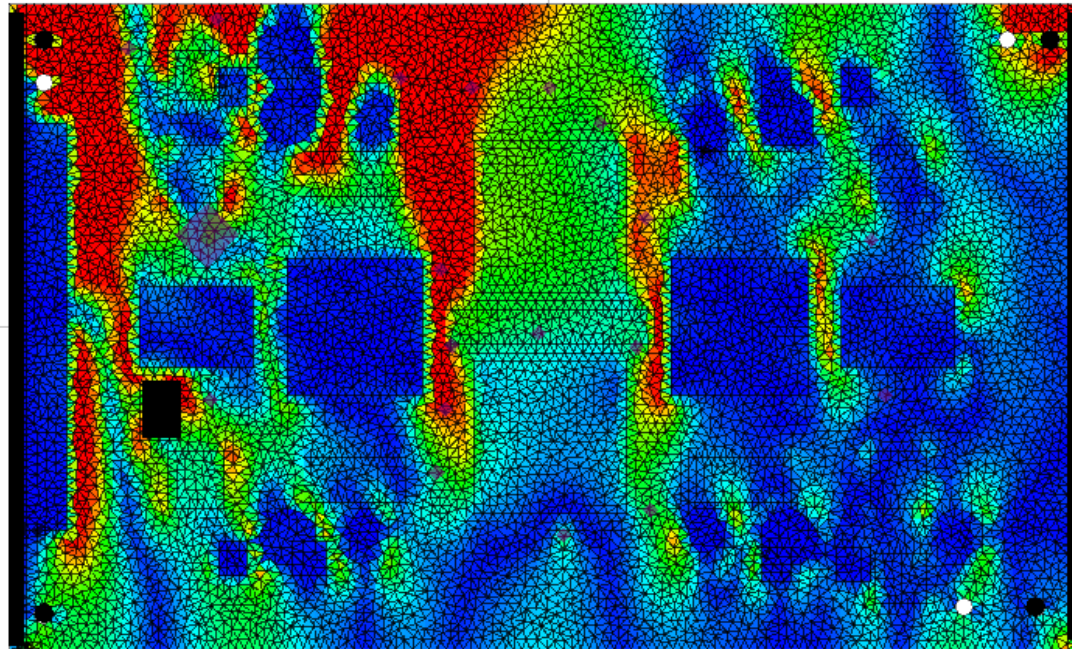
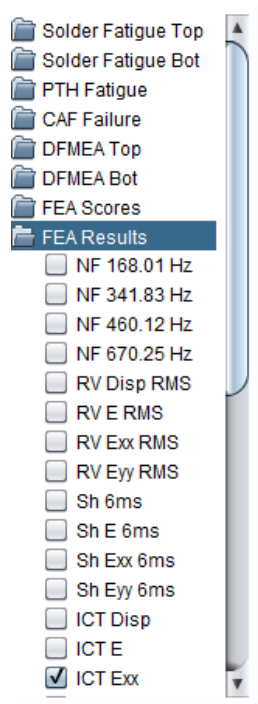
# 3D Output

The analysis can also establish 3D models by creating a mesh structure and the model from the data input to the analysis.





# In-Circuit Test Evaluation



- Uses embedded FEA engine to compute board deflection and strain cause by ICT fixture

# DFMEA

The screenshot displays a software interface for Design Structure Matrix (DSM) analysis. On the left is a 'Navigation' pane with a tree view. The tree is expanded to show 'U2' (indicated by a gear icon) and its sub-items: 'Logic Failure' (highlighted in blue), 'Open - Pin 1 (PLCC20-2)', 'Open - Pin 2 (N131610)', 'Open - Pin 3 (N131610)', 'Open - Pin 4 (N131614)', 'Open - Pin 5 (N131614)', 'Open - Pin 6 (N131618)', 'Open - Pin 7 (N131618)', and 'Open - Pin 8 (N131622)'. Above this are various other failure modes like 'Open - Pin 17 (N127243)', 'Short - Pins 1 (PLCC20-1) & 2', etc.

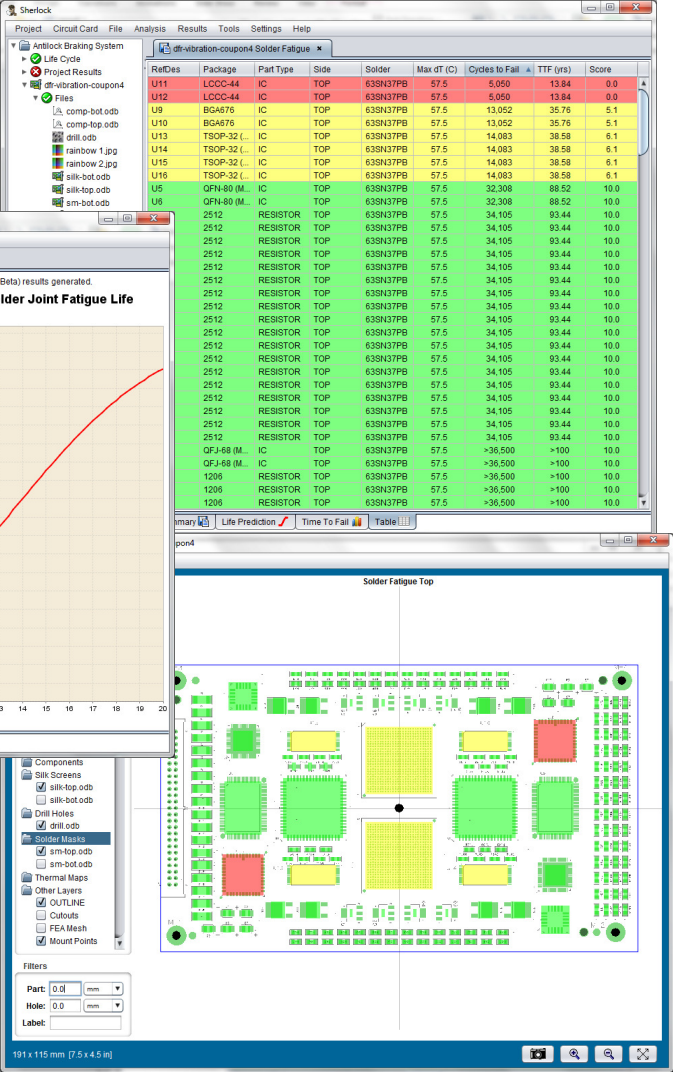
On the right, there are three stacked property panels:

- Subcircuit Properties:**
  - Subcircuit Name: UNASSIGNED
  - Description: (empty field)
- Part Properties:**
  - Part Designator: U2
  - Description: IC ADV PWM MOTOR CTRL LCCC-20
- Failure Mode Properties:**
  - Failure Mode: Logic Failure
  - Potential Cause: Device failure
  - Potential Effect: (empty field)
  - SEV: 8
  - OCC: 2
  - DET: 2
  - RPN: 32

- Uses ODB++ data including net list to create board level DFMEA
- Includes customizable spreadsheets for export

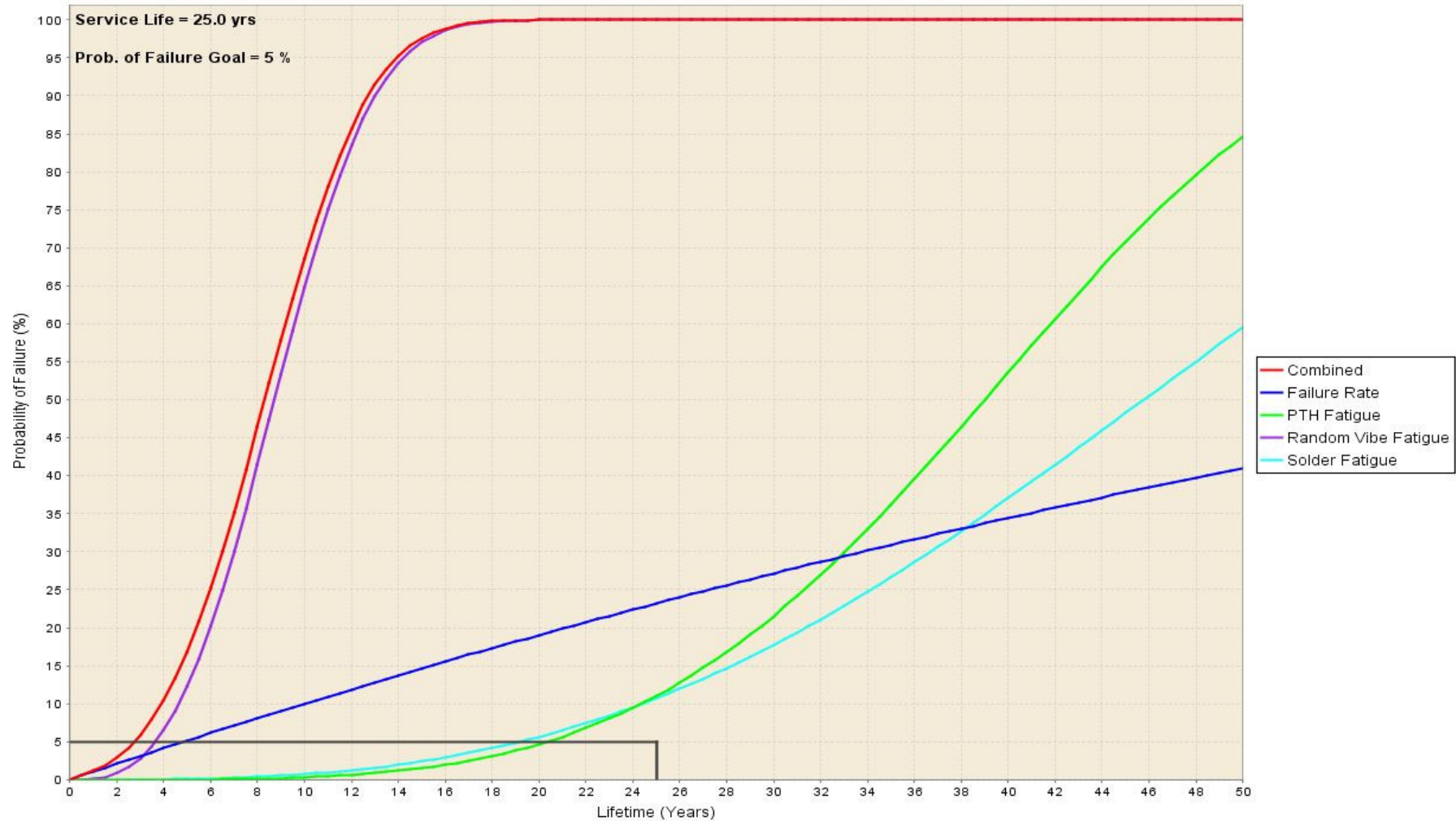


## Results: Five Different Outputs



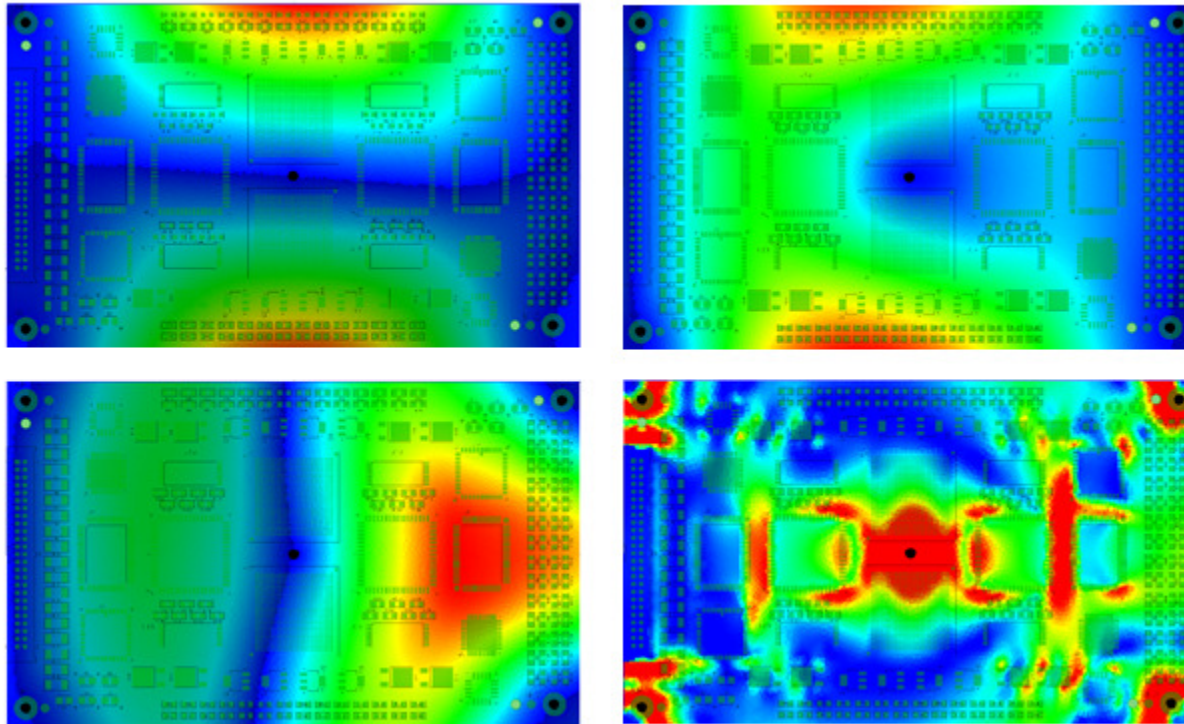
# Unreliability Curves

Tom's Demo ODB++ Tutorial - Life Prediction





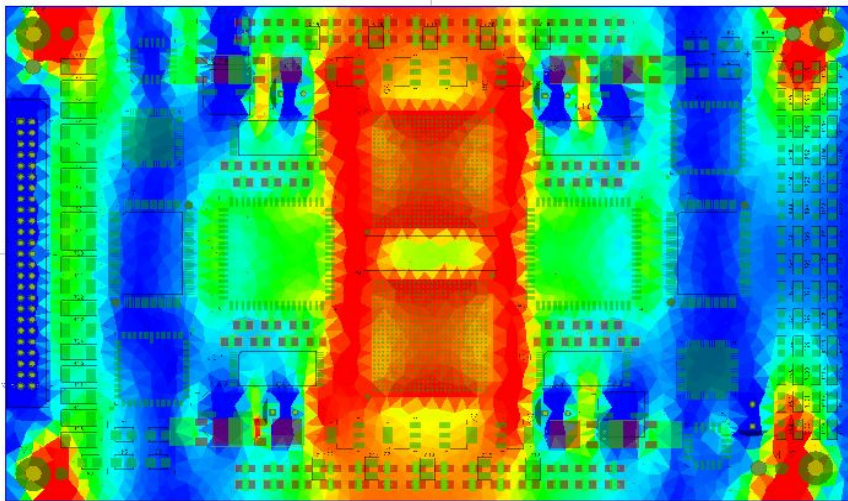
# Natural Frequencies



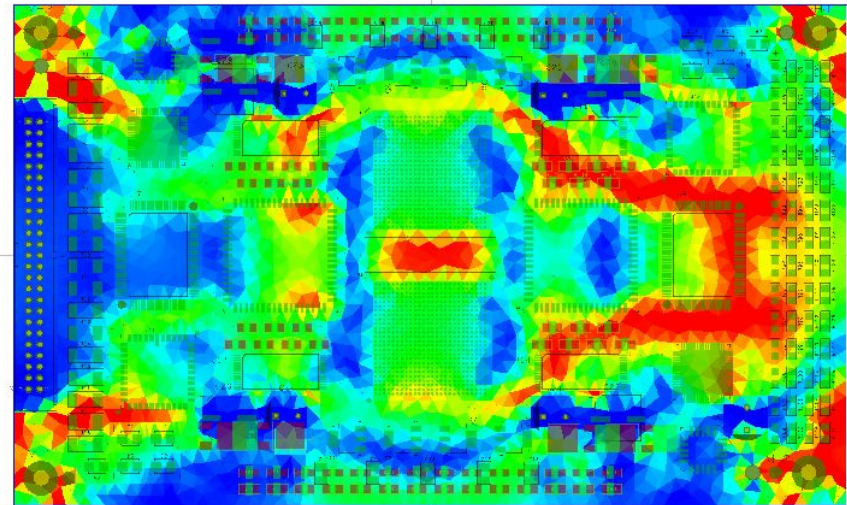
Natural Frequencies Identified (1<sup>st</sup>-upper left), (2<sup>nd</sup>-upper right), (3<sup>rd</sup> -lower left) and 4<sup>th</sup> – lower right)

# Vibration Strain Levels

Solder Fatigue Bottom / PTH Fatigue / Random Vibration RMS XX Strain / Failure Rate...



Solder Fatigue Bottom / PTH Fatigue / Random Vibration RMS YY Strain / Failure Rate ...

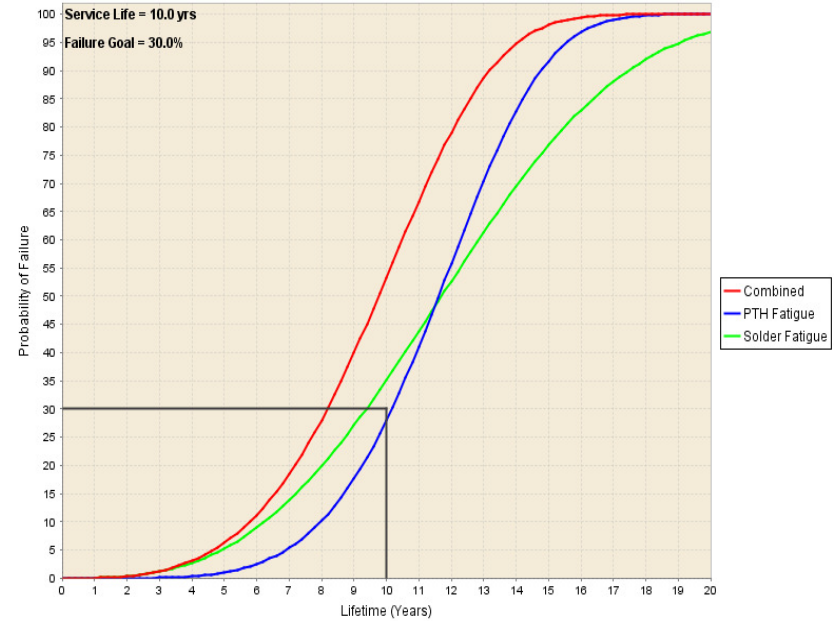
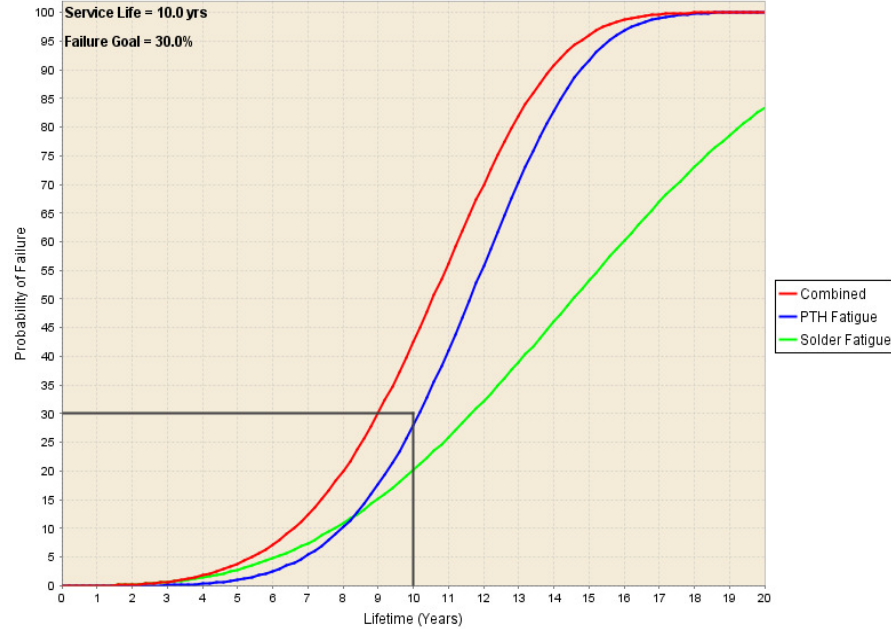


In addition, the analysis can provide data regarding the strains applied to the circuit board as a function of the vibration stress levels. The left illustrates this data in the XX direction and YY in the right image.



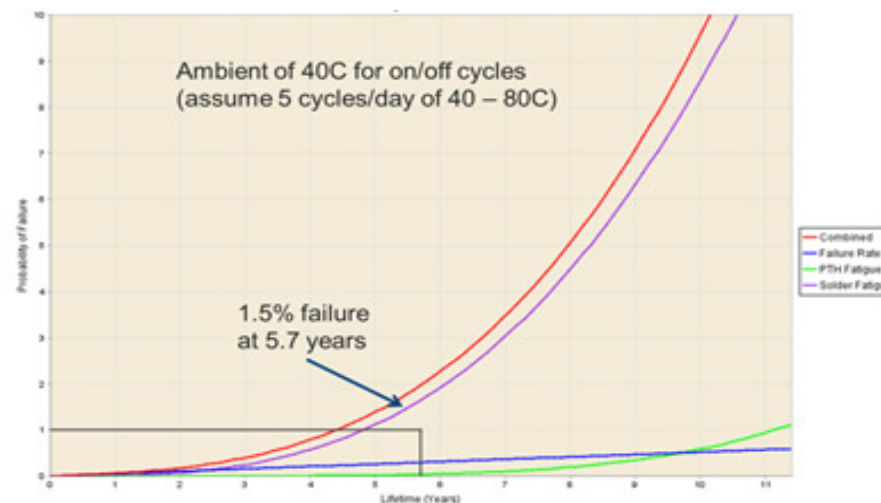
# What If?

Comparison of Sn/Pb (left) and SAC305 (right) with respect to solder fatigue



# Product Test Plans

- Product test plans, also known as design verification, product qualification, and accelerated life testing (though, these are not the same thing), are critical to the successful launch of a new product or new technology into the marketplace.
- These test plans require sufficient stresses to bring out real design deficiencies or defects, but not excessive levels that induce non-representative product failure.
- Tests must be rapid enough to meet tight schedules, but not so accelerated as to produce excessive stresses.
- Every test must provide value and must demonstrate correlation to the eventual use environment (which includes screening, storage, transportation/shipping, installation, and operation).





# DfR Solutions

reliability designed, reliability delivered

## Thank You!

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